

**MOON**  
VILLAGE ASSOCIATION



# AMPERE

**Autonomous Moon Power & Environmental Research Element**

**9th Global Moon Village Workshop & Symposium**

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MOON SURFACE IMAGE FROM: NICOLAS LEFAUDEUX (2021)

## Agenda

- ★ **Operating on the Moon: the energy gap**

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- ★ **Introducing AMPERE**

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- ★ **A 3-module architecture**

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- ★ **Eagle: Energy Generation & Storage**

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- ★ **Power Distribution and Demonstration**

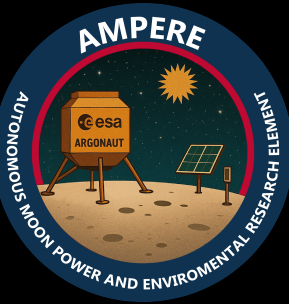
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- ★ **From AMPERE to a lunar power grid**

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- ★ **Conclusion and questions**

# Operating on the Moon: the energy gap



“Exploration”

**POWER  
DEFICIT**

“Utilization”

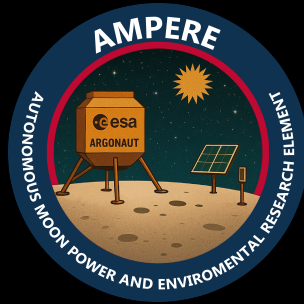
- ★ Industrial ISRU plants & construction are extremely power hungry: oxygen extraction plants require 30-70kW alone [1]
- ★ Lunar Night: 120 to 354 hours of darkness
- ★ Current model is not Scalable: every asset brings its own power

**Continuous power demand of  
20 - 100 kW [1]**

[1] J. ZHENG ET AL., "ELECTRICITY GENERATION FOR LUNAR BASES DURING CONSTRUCTION AND OPERATION: KEY TECHNOLOGIES, CHALLENGES AND DEVELOPMENT ROADMAP," ACTA ASTRONAUTICA.

## Introducing AMPERE

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# The first step toward a lunar power grid.

## What is it?

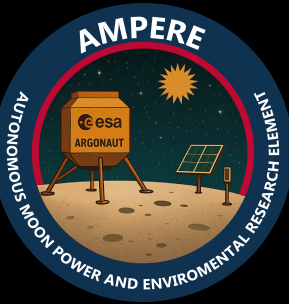
A demonstrative platform for power generation, storage and distribution on the lunar surface, while collecting environmental data

## Mission Objectives

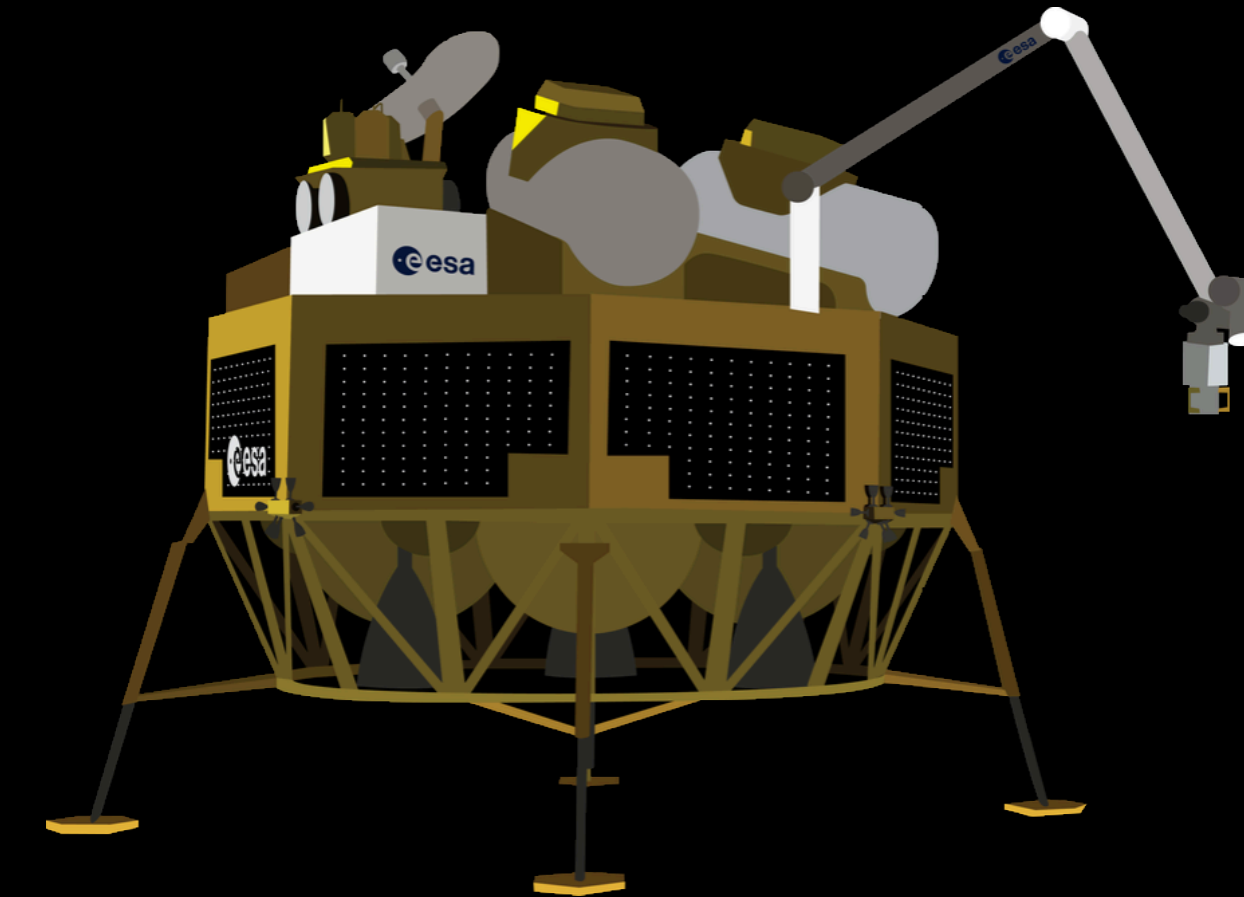
- ★ DEMONSTRATE THE GENERATION AND STORAGE AT SOUTH POLE  
**Generate and Store Energy**
- ★ DEMONSTRATE THE FEASIBILITY OF POWER DISTRIBUTION  
**Distribute Power**
- ★ VALIDATE LUNAR ENVIRONMENTAL MODELS  
**Research**

# Introducing AMPERE

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## The first step toward a lunar power grid.



ESA ARGONAUT LANDER. [2]

### European asset

- Ariane 6 launcher
- Argonaut lander
- Moonlight constellation

### Mission constraints

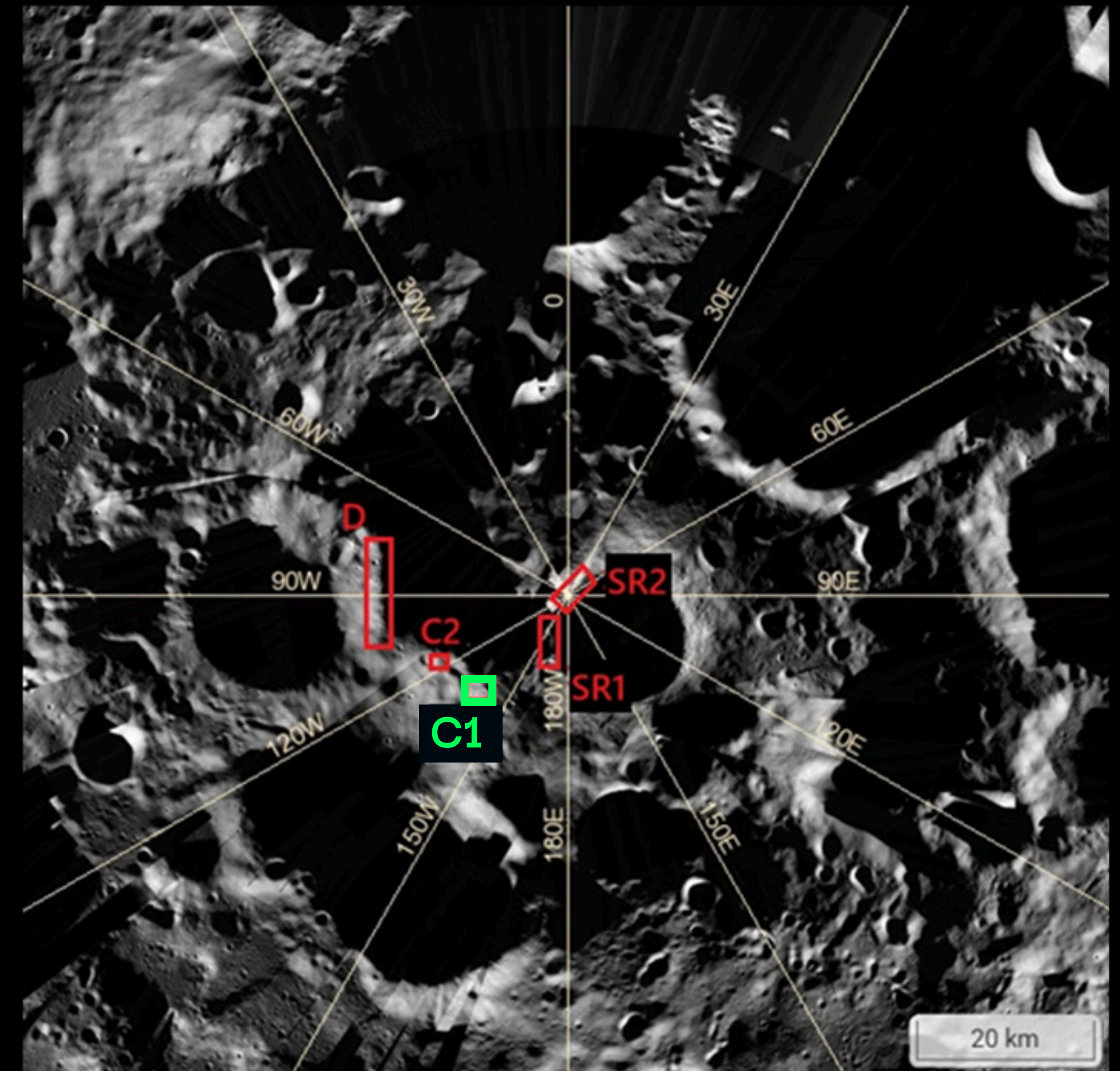
Mass	800 kg
Base surface	4 m <sup>2</sup>
Height	1.8 m

[2]: European Space Agency, "What is Argonaut?," infographic, ESA, Oct. 2020

# Introducing AMPERE

## Landing and Operational Site

- ★ Safe Landing Conditions
- ★ Critical Aspect for Payload Survival and Operativity
- ★ Average Illumination of 88% p.y. and 120 h of continuous darkness
- ★ Interest in Lunar South Pole



CANDIDATE LANDING SPOTS AT SOUTH POLE (RED)  
SELECTED LANDING SPOT CR1 (GREEN).[3]

[3]: NASA, "Moon Trek — A web portal for lunar data," Moon Trek. [Online]. Available: <https://trek.nasa.gov/moon/>

# A 3-Module Architecture

STATIONARY POWER HUB

## ★ EAGLE

- ~ 500 KG
- 5 KW OF GENERATED POWER
- 48 KWH OF STORAGE

DEMONSTRATIVE ROVER

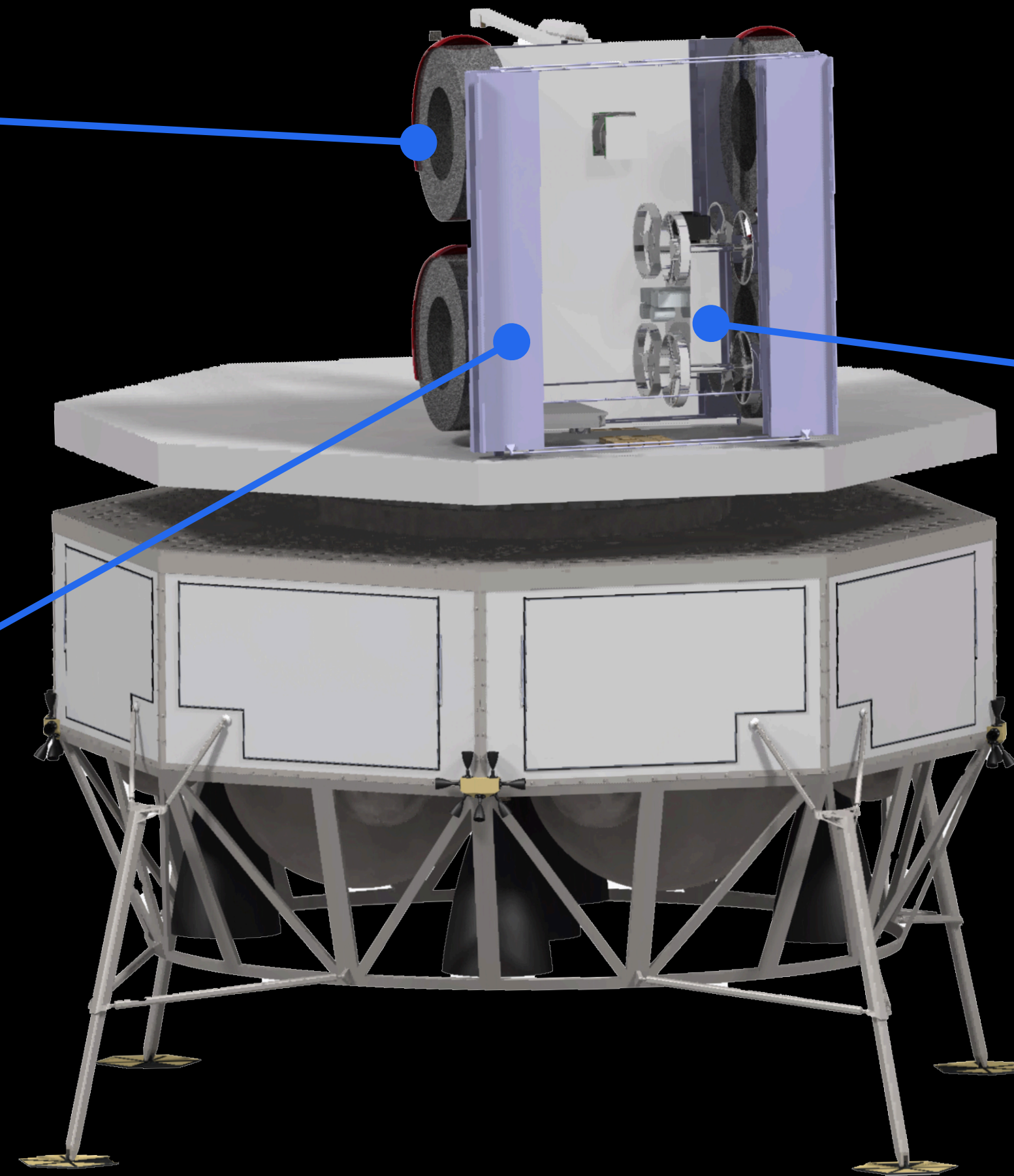
## ★ SCOTT

- 41 KG
- 297 WH BATTERY
- MOVES AT 6.5 CM/S
- 9 H DUTY CYCLE

PAYLOAD DELIVERY SYSTEM

## ★ PDS

- 6.86 M FOLDABLE RAMP @ 26° SLOPE
- 5 FOLDABLE PANELS
- 5 MECHANICAL HINGES
- ALUMINUM HONEYCOMB COVERED WITH CARBON FIBER REINFORCED POLYMER (CFRP) STRIPS

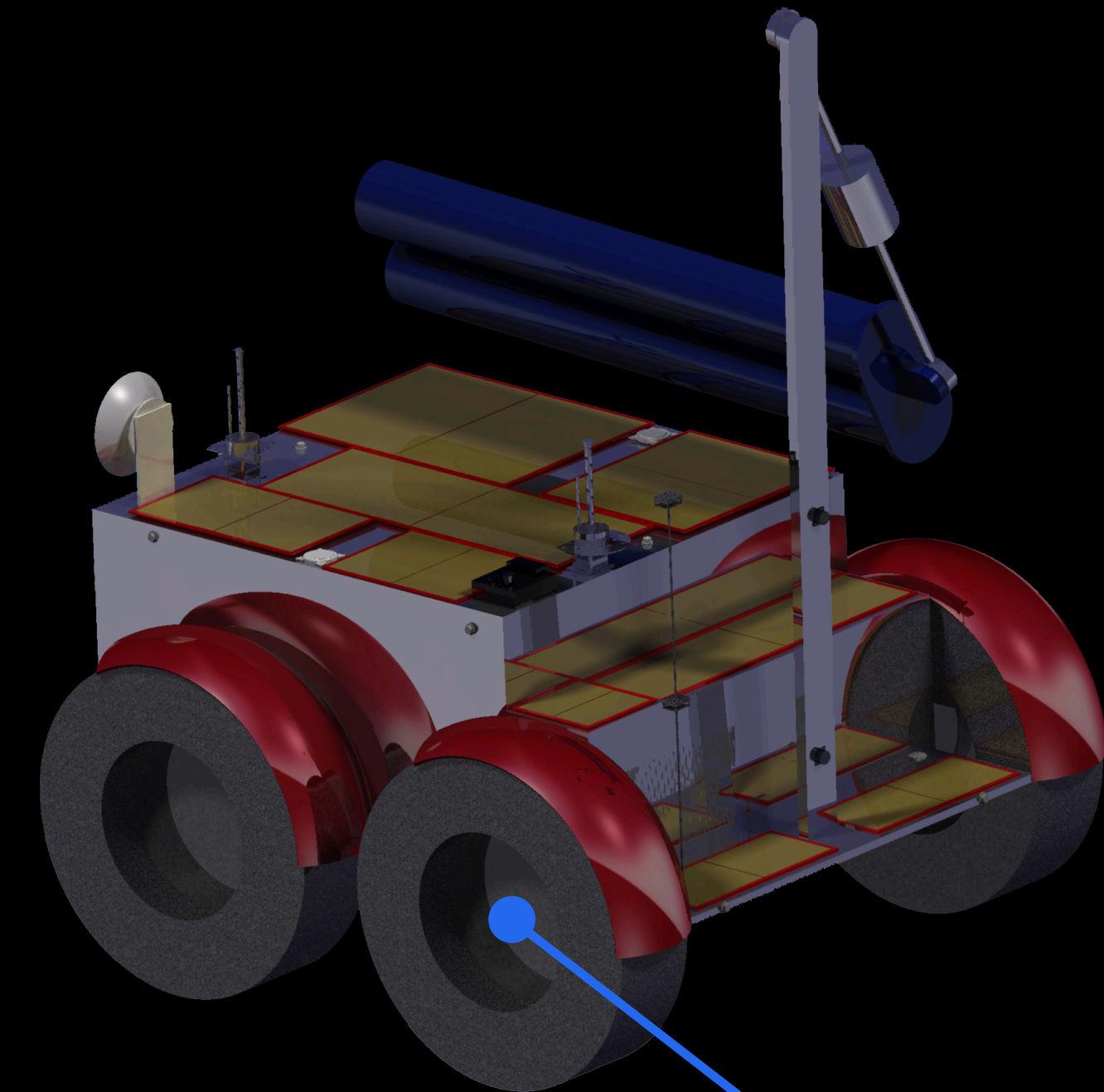


SCHEMATIC VIEW OF AMPERE ALLOCATED ON ARGONAUT

# EAGLE: Energy Generation and Storage

## Meet EAGLE

- Structurally built on honeycomb aluminum cart
- Dual-mission: generate and store energy to survive the 14-days of darkness



LUNAR PAYLOAD MODULE

★ **LPM**

LUNAR SENSOR MODULE - STATIONARY

★ **LSM-S**

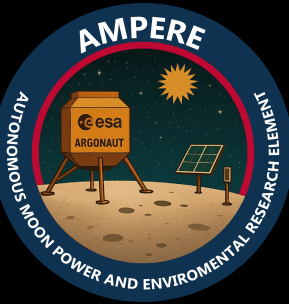
ROVER DELIVERY SYSTEM

★ **RDS**

WHEELS  
FOR DEPLOYMENT STABILITY  
AND POTENTIAL MICRO-POSITIONING

# EAGLE: Energy Generation and Storage

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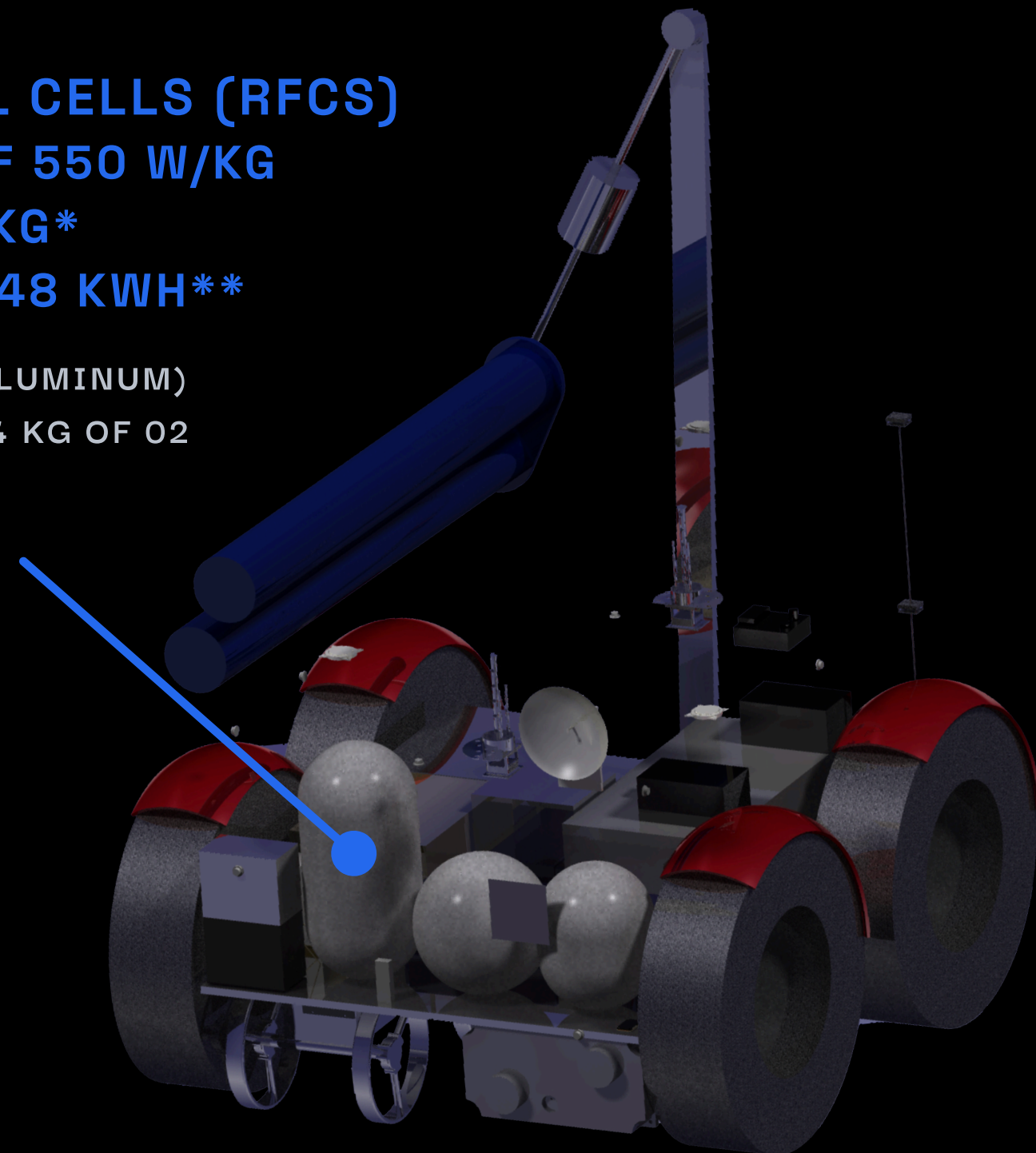
## Meet EAGLE

### REGENERATIVE FUEL CELLS (RFCS)

- POWER DENSITY OF 550 W/KG
- TOTAL MASS: 140 KG\*
- TOTAL CAPACITY: 48 KWH\*\*

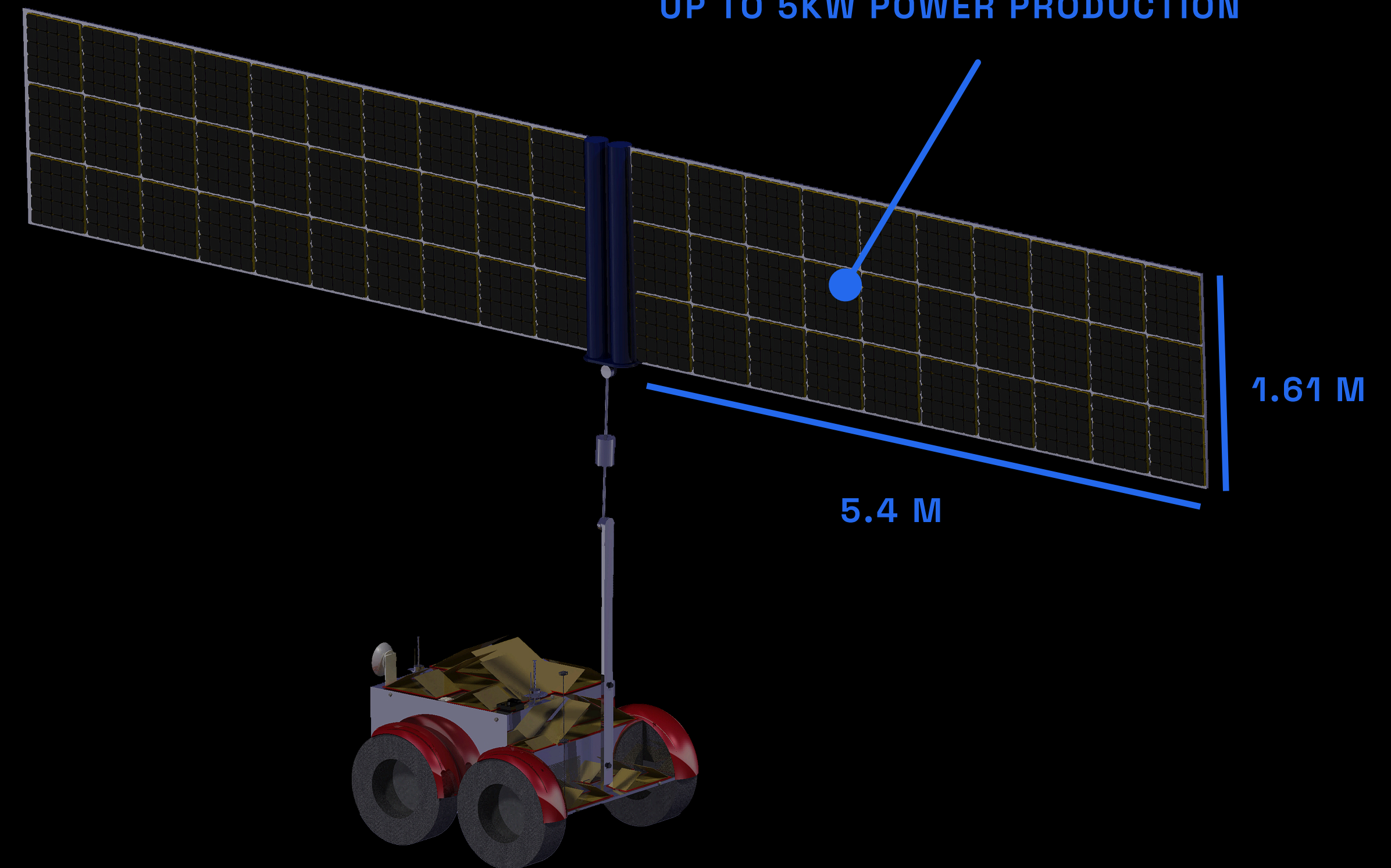
\*INCLUDING TANKS MASS (ALUMINUM)

\*\*USING 1.22 KG OF H<sub>2</sub> + 9.74 KG OF O<sub>2</sub>



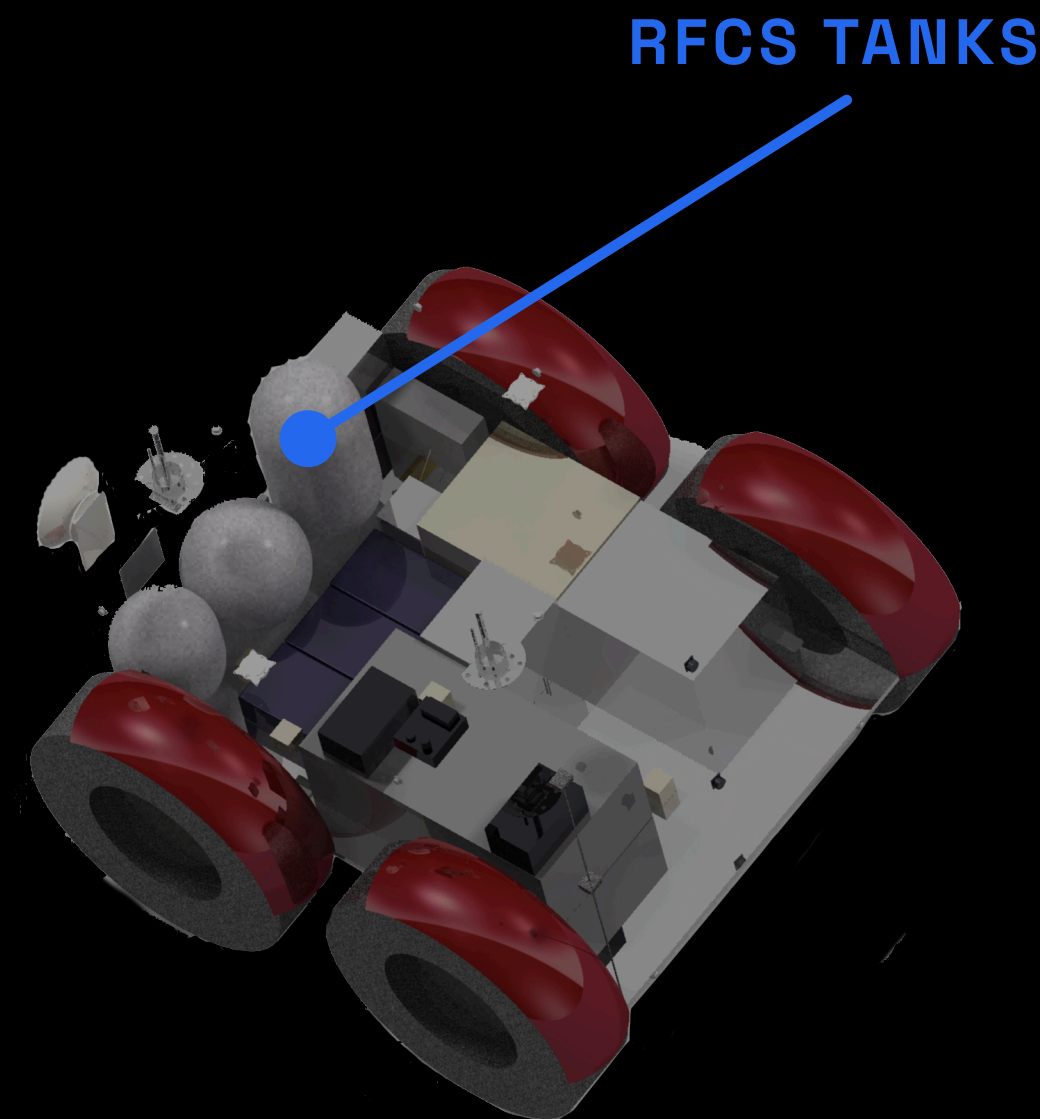
EAGLE INTERNAL COMPONENTS ARRANGEMENT  
(CLOSED PANELS CONFIGURATION)

ROLL-OUT SOLAR ARRAYS (ROSA)  
TOTAL SURFACE: 17.34 SQM  
QJ GAAS CELLS  
UP TO 5KW POWER PRODUCTION



EAGLE SOLAR PANELS DEPLOYED CONFIGURATION

## Meet EAGLE



EAGLE INTERNAL COMPONENTS.  
(UP-VIEW)



RADIATOR WITH TWO VANE  
BLADE LOUVERS [3]

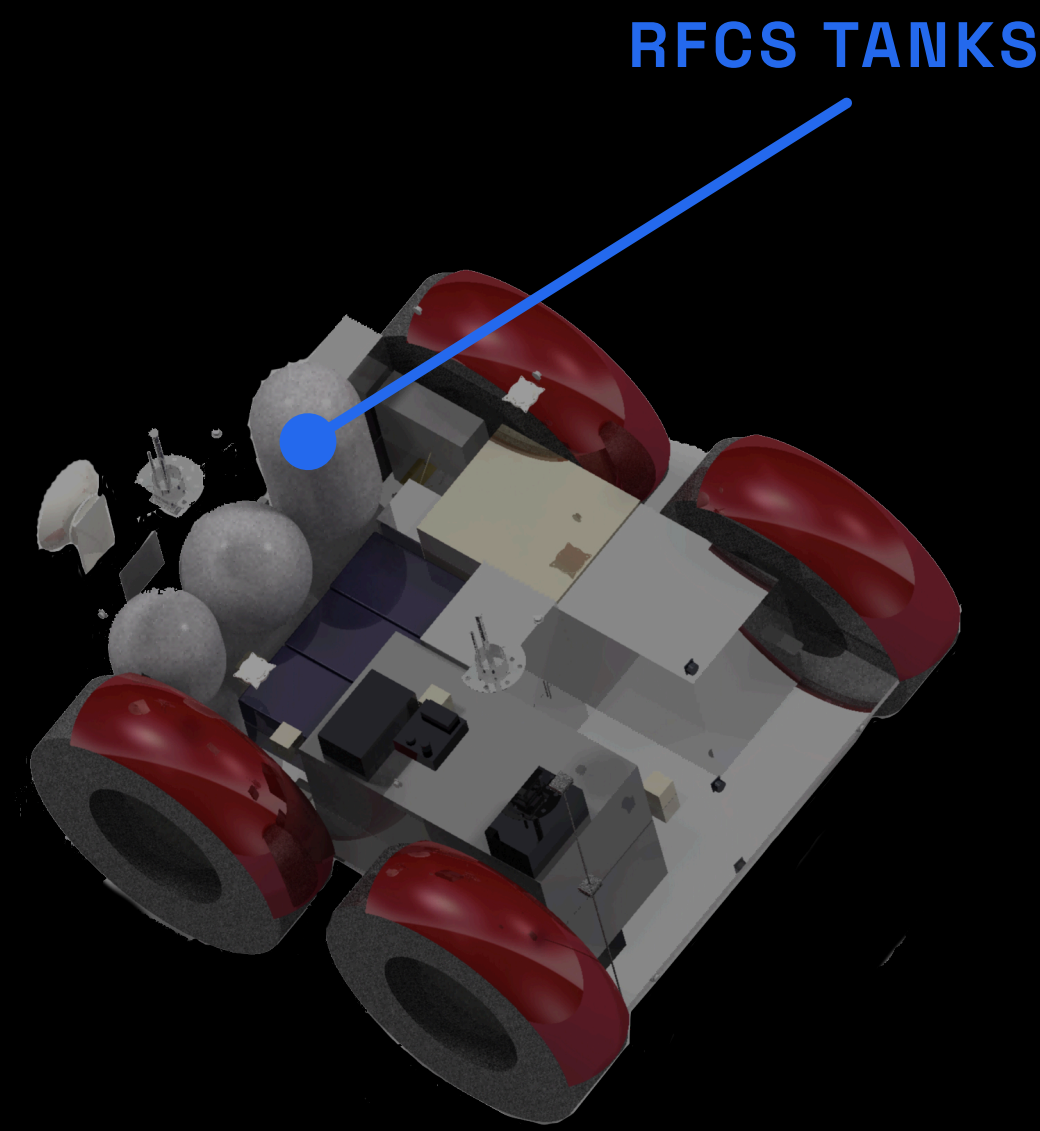
## Thermal Control System

Operative range: 273 K and 320 K

- | RFCS thermal synergy:  
at 50% electrical efficiency they allow to recover waste heat
- | Smart Multi-Layer Insulation (SMLI) with variable emissivity and Loop Heat Pipes (LHP)  
**POWER CONSUMED: 108.6 W**
- | Passive radiators plus blade louvers dissipate heat during daytime

[3]: VANOVA, Dmitri; FERNANDES, Domingos. Lunar dust tolerant thermal louver development for a light-weight rover system. Lunar dust tolerant thermal louver development for a light-weight rover system

## Meet EAGLE



EAGLE INTERNAL COMPONENTS.  
(UP-VIEW)



RADIATOR WITH TWO VANE  
BLADE LOUVERS [3]

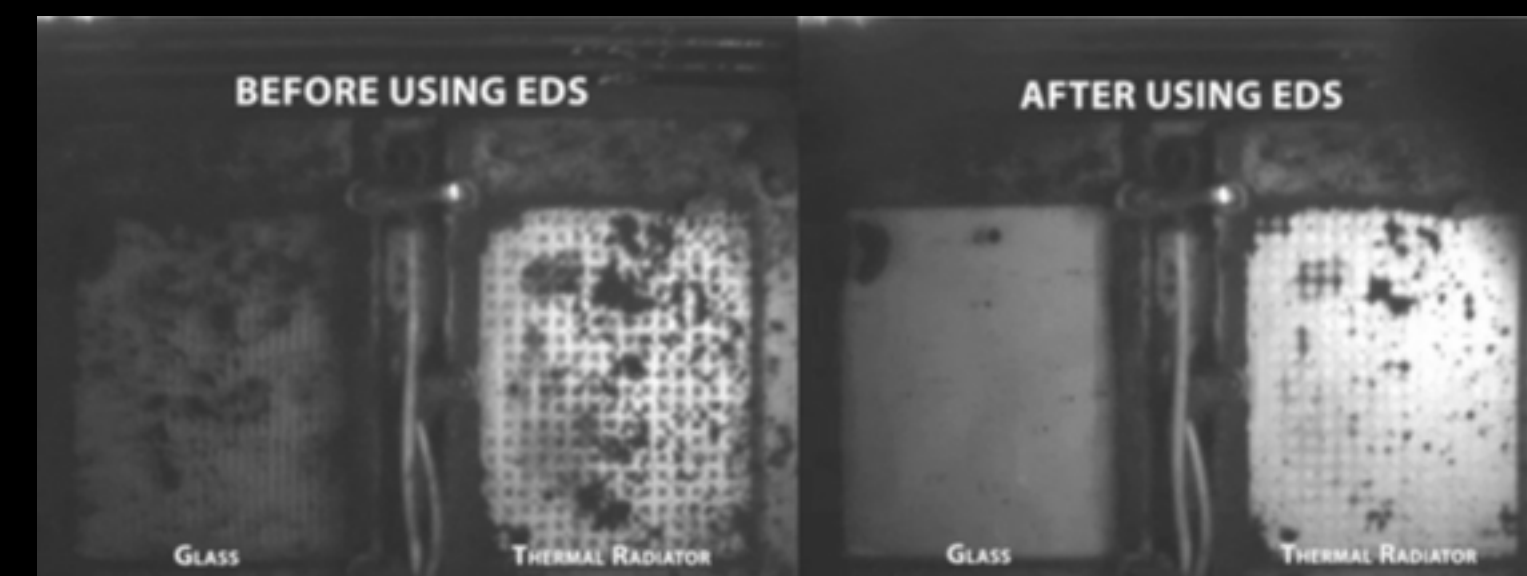
## Thermal Control System

Hybrid thermal control

### ACTIVE DUST MITIGATION:

Electro-dynamic Dust Shield (EDS) prevents regolith accumulation on radiators and MLI to maintain thermo-optical properties

POWER CONSUMED: 49.5 W



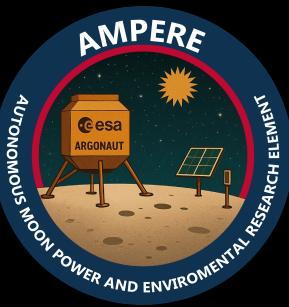
EDS EFFECT ON GLASS AND THERMAL RADIATORS.

[3]: VANOVA, Dmitri; FERNANDES, Domingos. Lunar dust tolerant thermal louver development for a light-weight rover system. Lunar dust tolerant thermal louver development for a light-weight rover system

[4]: NASA. NASA's Dust Shield Successfully Repels Lunar Regolith on Moon [online]. NASA, 2024. Available: <https://www.nasa.gov/image-article/nasas-dust-shield-successfully-repels-lunar-regolith-on-moon/>

# EAGLE: Energy Generation and Storage

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LUNAR PAYLOAD MODULE

★ LPM

LUNAR SENSOR MODULE - STATIONARY

★ LSM-S

ROVER DELIVERY SYSTEM

★ RDS

## Sensors on-board

To evaluate environment and elimination of movement:

### Measurements:

- Total ionizing dose
- Radiation (with NGRM\*)
- 3-axis fluxgate magnetometer
- PRT (Platinum Resistance Thermometer)

\*NEW GENERATION RADIATION MONITOR

## How does EAGLE transfer power?

Two standards are selected:

WIRELESS

### ★ UFPC

125 W @ 8A

VIA INDUCTIVE COUPLING.

NO EXPOSED CONTACTS, NO DUST RISK



FROM LEFT TO RIGHT: TRANSMITTER, TRANSMITTER COIL, ON BOARD CHARGER, RECEIVER COIL [5]

WIRED

### ★ LESSH

215 W @ 9.4 A

WIRED INTERFACE

FOR FUTURE HEAVY ASSETS



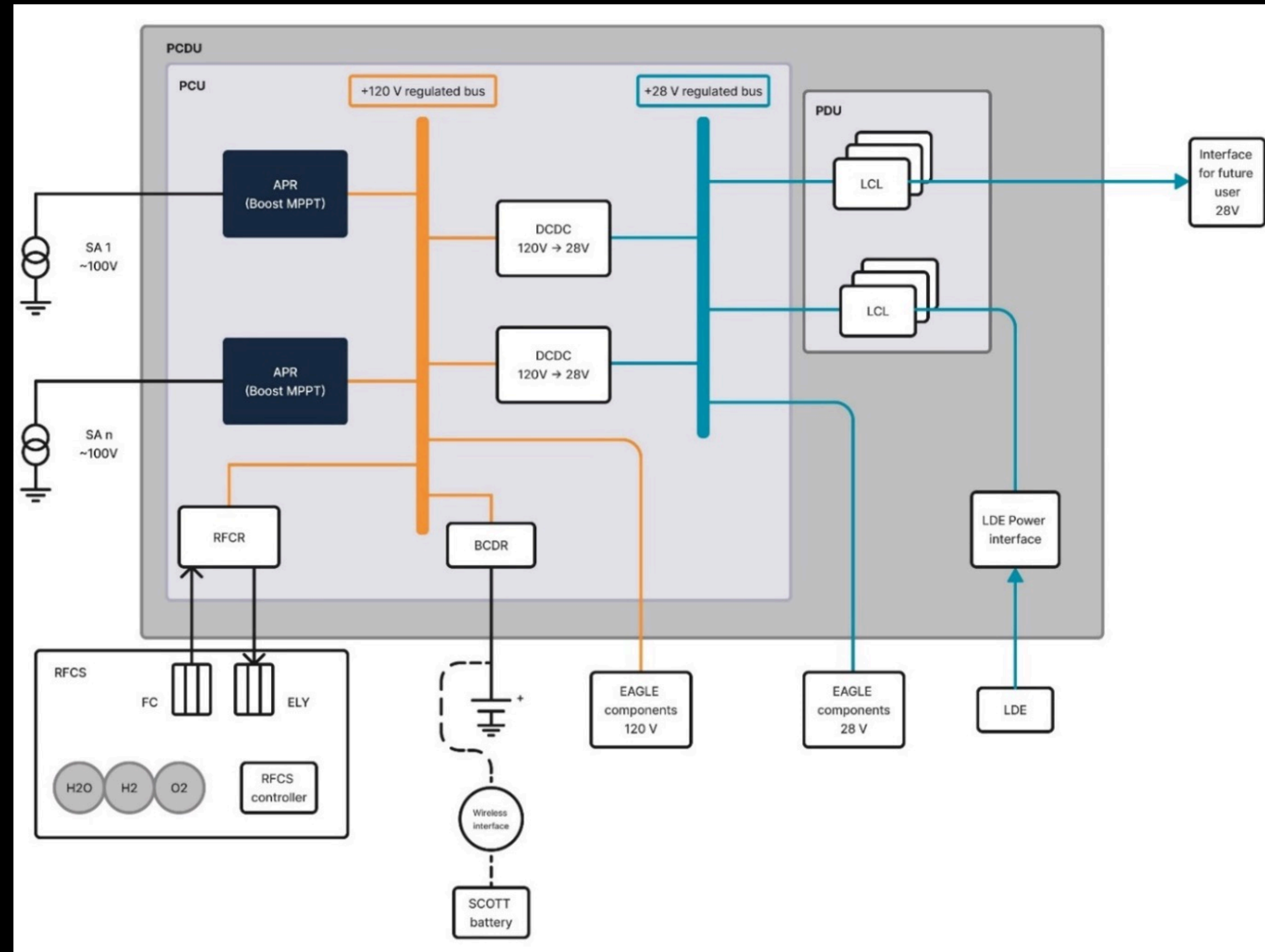
LESSH BCM [6]

[5]: Ultra Fast Proximity Charger: Wireless Power for the Moon,” Astrobotic, Pittsburgh, PA, Nov. 2023. [Online]. Available: <https://science.nasa.gov/wp-content/uploads/2023/11/astrobotic-wireless-charger.pdf>

[6]: Amato, Michael, et al. “LESSH Lunar Experiment Support System and Handling Battery Charger Module.” Lunar Surface Innovation Consortium. 2024.

# EAGLE: Energy Generation and Storage

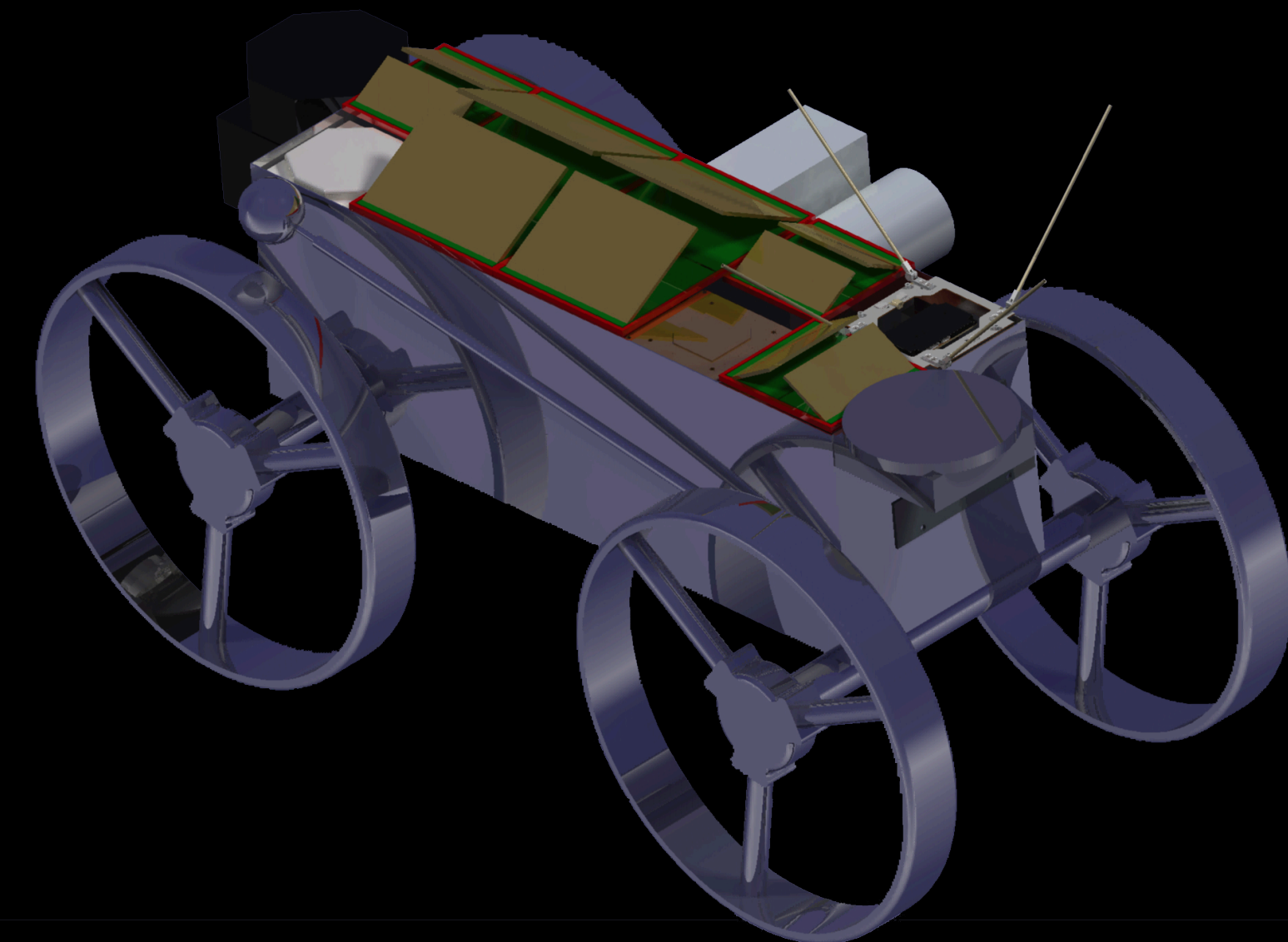
- EAGLE's PCDU scheme fully integrated with LDE
- 1.6 kW are used to sustain the platform during daytime
- 3.4 kW available to charge other asset
- Future assets will use the 28V bus, while SCOTT uses the 120V



EAGLE PCDU SCHEME

# Scott: Demonstrative Rover acting as future customer

- | 24U CubeRover equipped with a 297Wh battery
- | Capable of moving at 6.5 cm/s



SCOTT ROVER.  
EXTERNAL VIEW

## Primary job

Traverse 50 to 75 m, perform scientific readings and return to Eagle to recharge

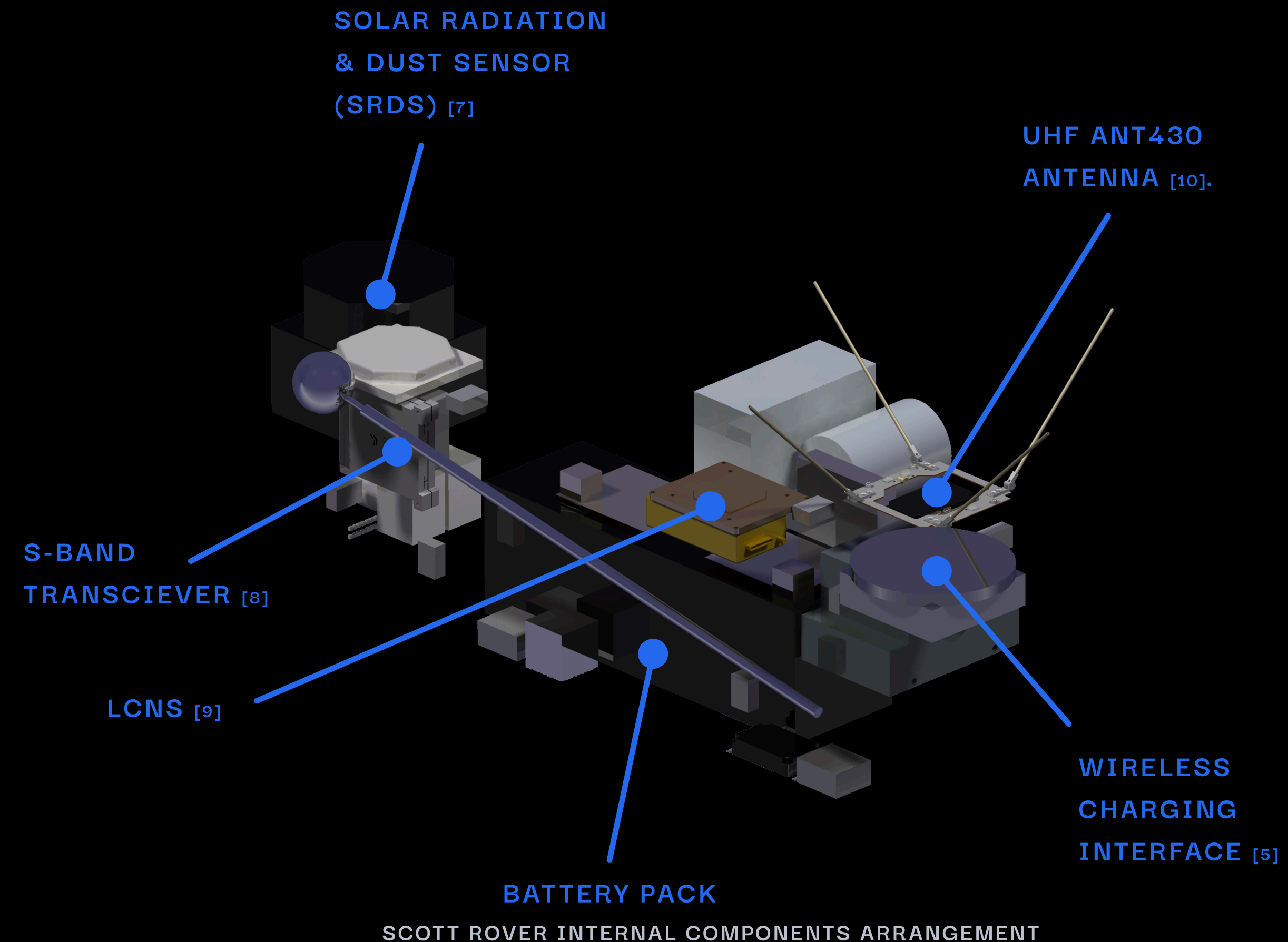
## Scott: Demonstrative Rover acting as future customer

Operational cycle of 9 hours:

- 2 h for complete charging
- 7 h for moving and data acquisition

Equipped with:

- Dosimeter
- Radiation monitor
- Temperature sensor
- 3-axis magnetometer



[7]: UJ. A. Rodriguez-Manfredi et al., "The Mars Environmental Dynamics Analyzer (MEDA): A suite of environmental sensors for the Mars 2020 mission," \*Space Science Reviews\*, vol. 217, no. 48, 2021

[8]: EnduroSat, "S-Band Transceiver for CubeSat and SmallSat Communication," EnduroSat. [Online]. Available: <https://www.endurosat.com/products/s-band-transceiver/>

[9]: SpaceQuest, Ltd., "GNSS-700 Series: Satellite GNSS Receiver," ver 2019-08-09, Fairfax, VA, USA, Aug. 9, 2019. [Online]. Available: [\[https://www.aac-clyde.space/product/gnss-700-series\]](https://www.aac-clyde.space/product/gnss-700-series)

[10]: GomSpace, "NanoCom ANT430 — Wire Antenna," SatCatalog (ora SatBase). [Online]. Available: <https://www.satcatalog.com/component/nanocom-ant430/>

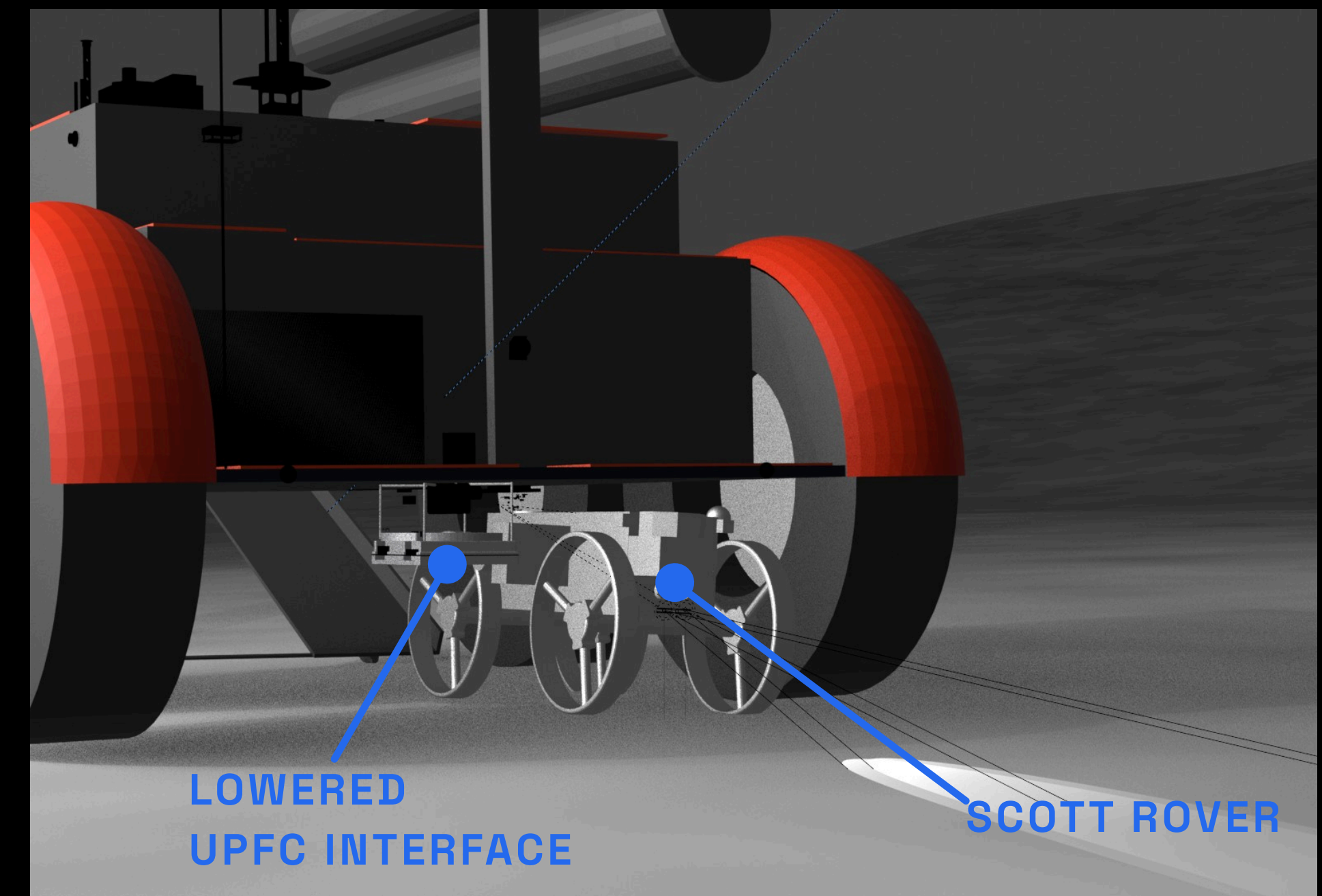
# Scott: Demonstrative Rover acting as future customer

Released on the Moon surface thank to RDS:

- located underside EAGLE's chassis
- uses Ejector Release Mechanism (ERM) to drop Scott (10 cm, absorbed by wheels)

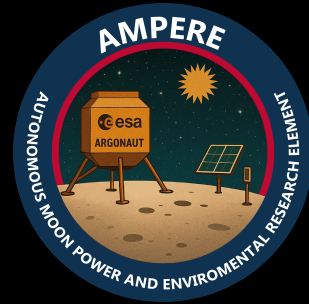
RDS also:

- Lowers the Wireless Power Transmitter from Eagle to a precise height of 35 cm a.g.



RENDER OF LOWERED UPFC AND SCOTT.

From AMPERE to a lunar power grid.



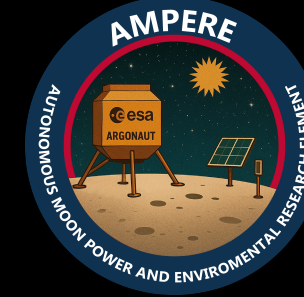
## Not a one-off mission.

AMPERE is designed to survive 12 months and validate the technology by setting a standard for power distribution.

Other assets can use EAGLE to charge during the mission period

EAGLE is a single block. More blocks can be used to form a Micro Power Grid and meet the energy demand

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**Lowering the barrier for lunar surface**

**Thank You**

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AMPERE mission